

# Why Projects at iacmi are Important to Airbus

IACMI Fall Member Meeting 2020

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**AIRBUS**

# Agenda

- Zero Emission Commercial Aircraft
  - What it is
  - Rough timeline
- iacmi 5.4 Injection Overmolding of Continuous Carbon Fiber Preforms
- iacmi 5.5 Hybrid Additive Manufacturing (AM) Tooling for Large Composite Structures
- iacmi 5.7 Tailored Fiber Placement for Complex Preforms

# Zero-emission Commercial Aircraft by 2035



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# Zero-emission Commercial Aircraft by 2035

- Next 5 years identify technologies
- Then a couple of years to determine industrial, supply chain and manufacturing processes to support production
- 2027-2028
  - Begin detailed design of new H2 powered aircraft
- Goal: Mid 2030's
  - ZEROe aircraft entered into service

## 5.4 Injection Overmolding of Continuous Carbon Fiber Preforms

### Objective

This project proposes utilizing continuous carbon fiber reinforcement to provide the structural strength and stiffness needed, but also combining it with the rapid manufacturing and lower costs/cycle times achieved by injection molding.

### Partners

Airbus, UDRI, Zoltek, SABIC, JobsOhio

### Technical Contribution to Industry

The project provides weight, energy, and cost savings as the TFP, combined with injection overmolding, allows the design to be optimized to suit the actual loading conditions.

### Injection overmolding importance to Airbus

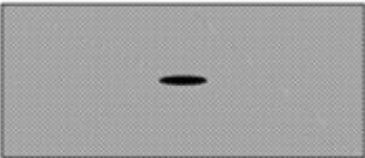
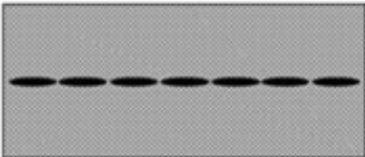
- Weight reduction by metal replacement
- Cost reduction
- Galvanic corrosion protection by replacing metal with GFRP
- Contribute to sustainability

# 5.4 Highlights

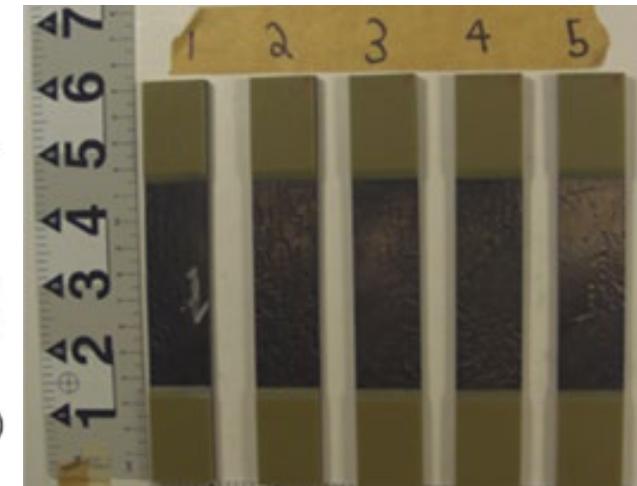
## Flat Panel Results

- Tailored Fiber Placement (TFP) reinforced panels have a greater than 25% increase in tensile strength compared to short fiber reinforced thermoplastic specimens

## Flat Panel Tensile Strength

	EC004APQ Carbon Fiber filled PEI	Finite Element Model Prediction: 111 MPa Test Result: 84.5 MPa
	EC004APQ Carbon Fiber filled PEI Single 12k Carbon Fiber Tow	Finite Element Model Prediction: 117.5 MPa Test Result: 105 Mpa (24% over baseline) 11.0% Difference between modeled/tested
	EC004APQ Carbon Fiber filled PEI 7 12k Carbon Fiber Tows (20% VF)	Finite Element Model Prediction: 205 MPa Test Result: 194 Mpa (130% over baseline) 5.5% Difference between modeled/tested

Tensile specimens cut from panels

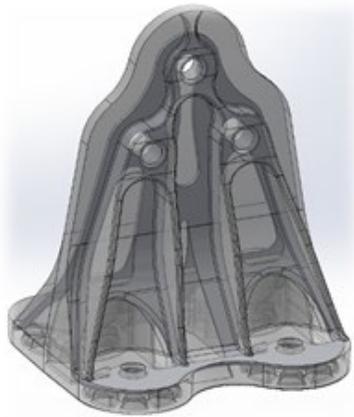


## 5.4 Ongoing Work

### Molded Part Results

- TFP preform and injection molding tooling was designed based on Airbus part geometry.
- Results to be compared to with aluminum and chopped fiber reinforced technology

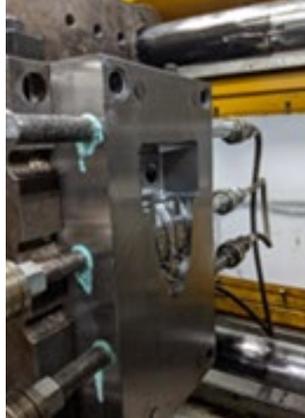
Design



TFP preform



Injection Mold



Structural Testing



Molding trials with different preform designs are being tested.



## 5.5 Hybrid (Additive Manufacturing) AM Tooling for Large Composite Structures

### Objective

Develop a hybrid AM tool capable of producing a 10 foot composite part. The tooling must be able to withstand multiple cycles in a 350F, 90psi autoclave and exhibit an isotropic thermal expansion matching that of Invar.

### Partners

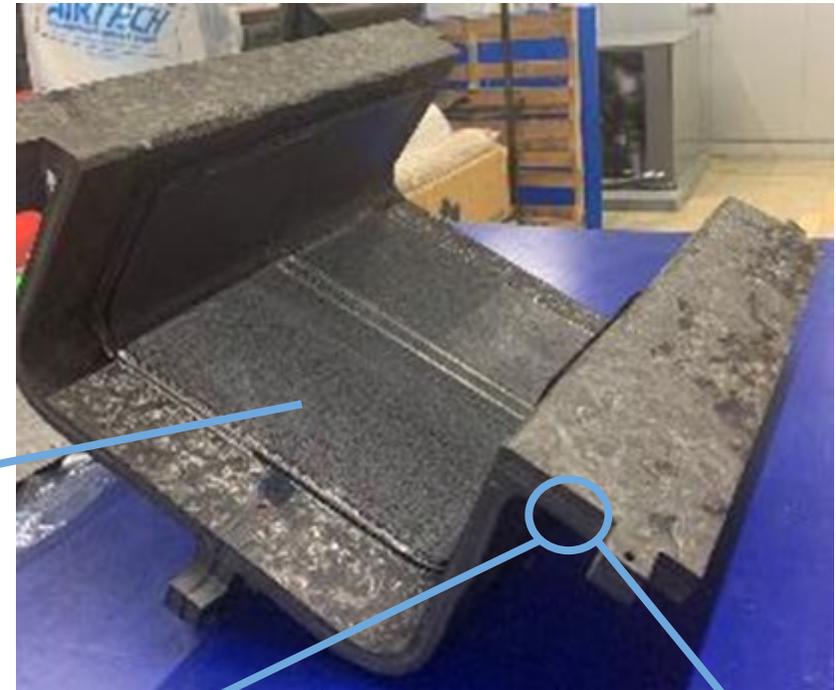
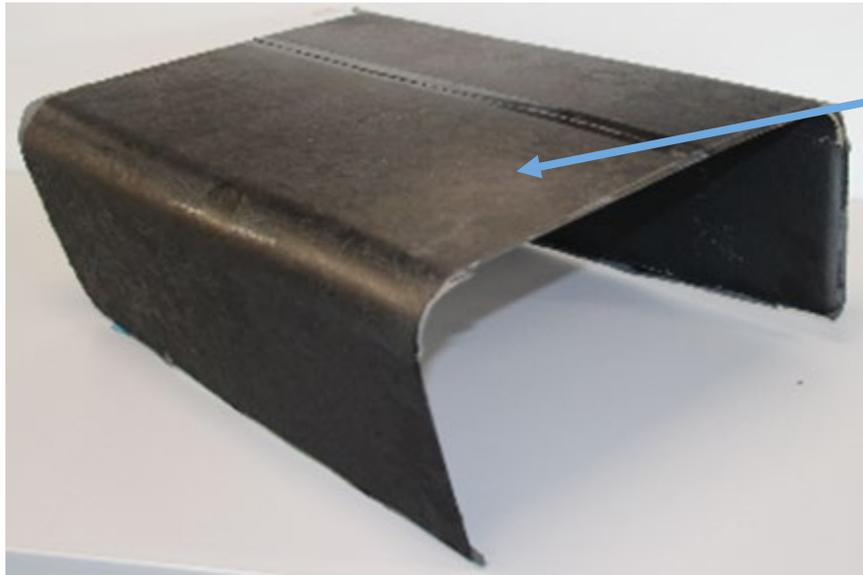
Airbus, Northrop Grumman, CI, AES, UDRI, JobsOhio, supported by SABIC and Hexcel

### Technical Contribution to Industry

Use AM to provide a low to mid rate production tool at 50% cost while matching thermal performance of Invar.

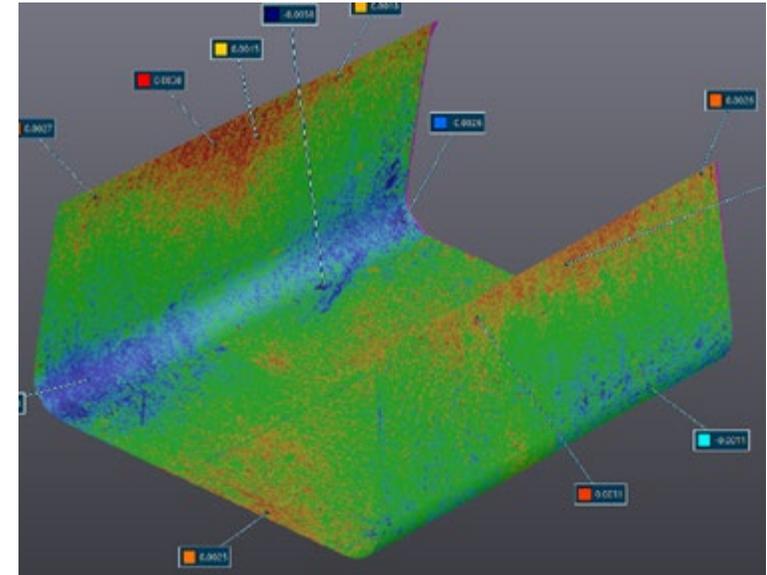
## 5.5 Highlights

- BAAM-printed core to near net shape dimension (Airbus C-spar)
- Overlaid Hextool™ tooling prepreg on as-printed surface
- Final machining of prepreg surface

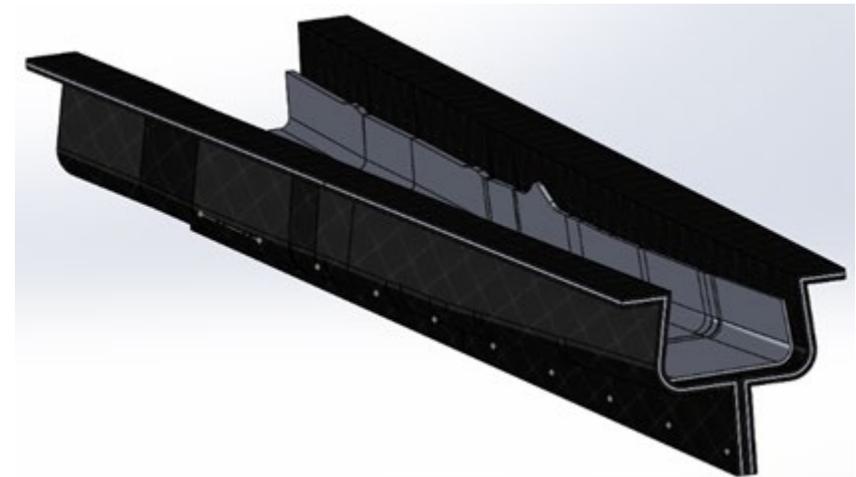


# 5.5 Results

- Nearly isotropic CTE within a few ppm/deg C from Invar
- Ten autoclave cycles with no tool damage
- Less than +/-0.005” tool deflection between Part 1 (Cycle 1) and Part 2 (Cycle 11)
- Tool cost \$24,200 versus \$46,800 for identical Invar tool (confirmed with external commercial quotes)



		PEI Tool (as-printed)	PEI Tool (with 5 plies HexTool)
CLTE <sub>177C</sub> ppm/C	flow	5.8	2.6
	x-flow	57.1 →	6.3



## 5.7 Tailored Fiber Placement for Complex Preforms

### Objective

Develop the confidence of computational tools to design and build TFP primary structures, that have an optimized fiber architecture integrating metallic features to improve performance while reducing costs/weight.

### Partners

Airbus, Lockheed Martin, MSU, ZSK, UDRI, JobsOhio

### Technical Contribution to Industry

Develop and demonstrate analytical tools are capable of predicting TFP component performance in order to optimize a part design to improve performance against traditionally fabricated components.

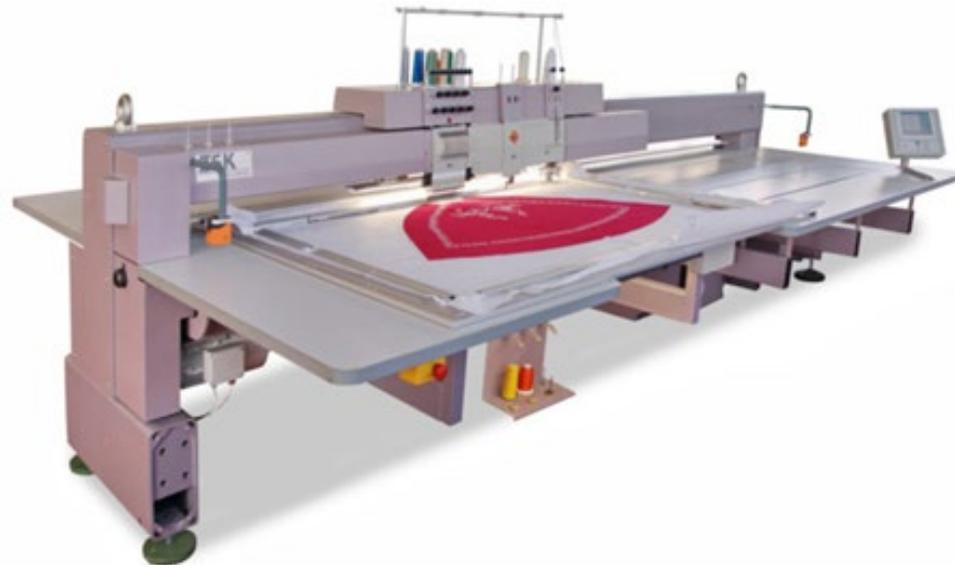
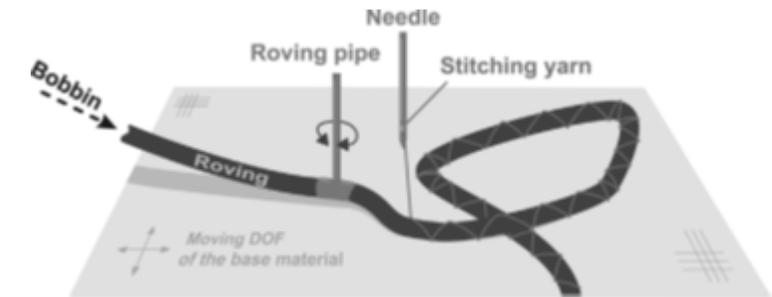
### TFP importance to Airbus

➤ Allow integration of reinforcements for highly loaded structure

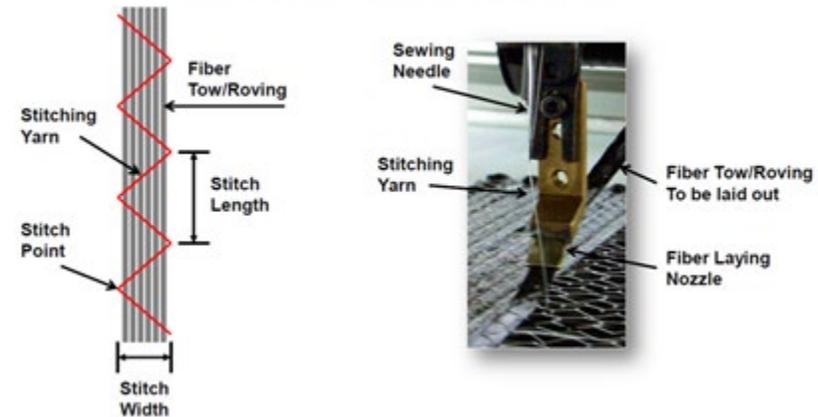
# 5.7 Tailored Fiber Placement

## Tailored Fiber Placement

- Embroidery-based technology
- Automated tow steering process known as Tailored Fiber Placement (TFP)



## Principle of TFP Process



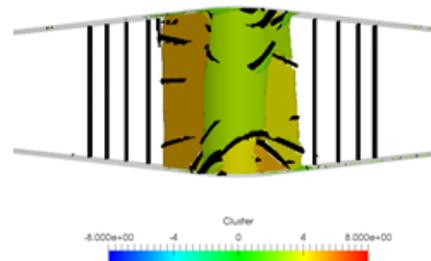
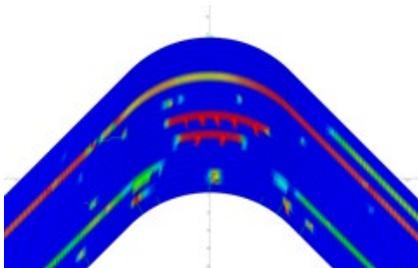
## 5.7 Highlights

- Characterized mechanical performance and distortions as function of stitch density
- Showed z-stitching can greatly increase strength and toughness
- Developed computational methods of simulating various fiber and stitch architectures



Mode I testing of  
Stitched TFP  
preform

Damage Modeling in Curved TFP Beams

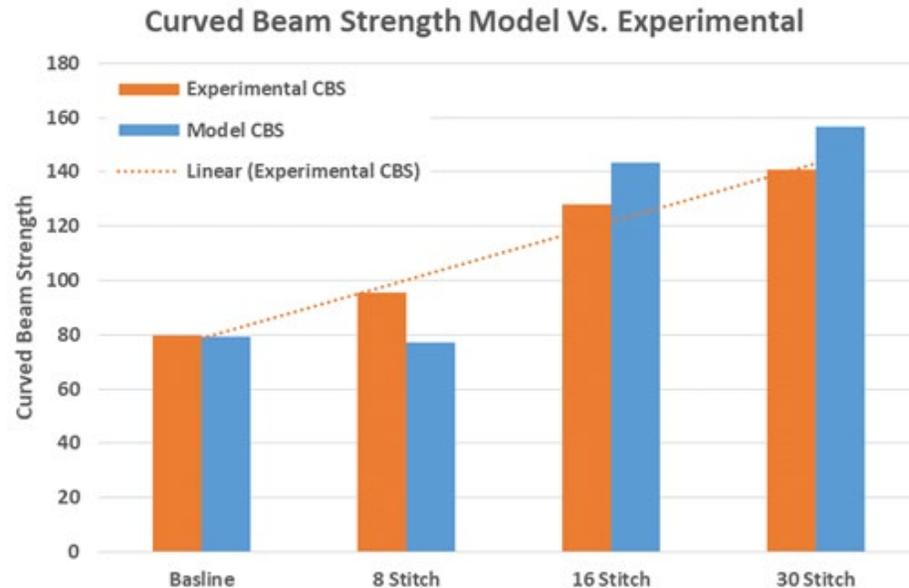


Damage Observations in TFP Curved Beams



## 5.7 Results

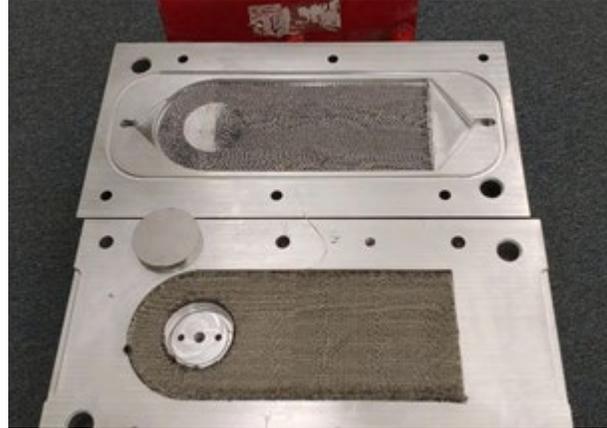
- Verified that stitching leads to large increase in Mode I toughness and ILT
- Model captures increase in Curved Beam Strength as a function of stitch density (model within 18%)
- Model can capture fiber distortions leading to changes in performance



	Fracture Toughness (kJ/m <sup>2</sup> )	% Improvement G <sub>IC</sub>	ILT (MPa)	% Improvement ILT
Baseline	0.445		19.54	
8 Stitches/cm <sup>2</sup>	1.93	333.71	22.99	17.66
16 Stitches/cm <sup>2</sup>	4.22	848.31	26.3	34.60
30 Stitches/cm <sup>2</sup>	7.45	1574.16	32.34	65.51

## 5.7 Ongoing Work

- Building Primary Lug Structure
  - Computational model running for blind predictions

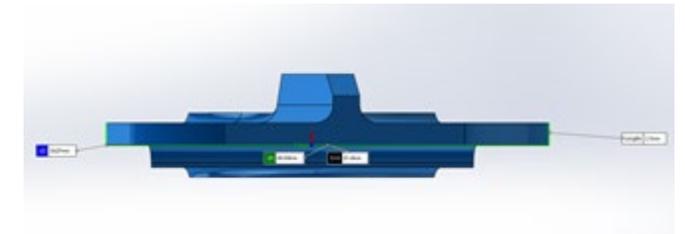
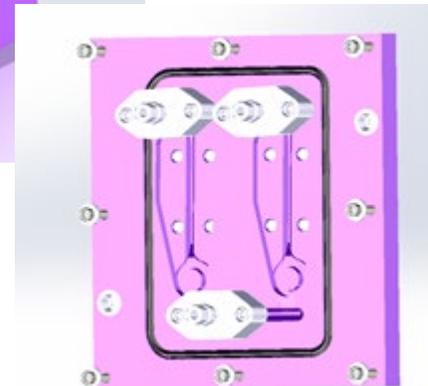
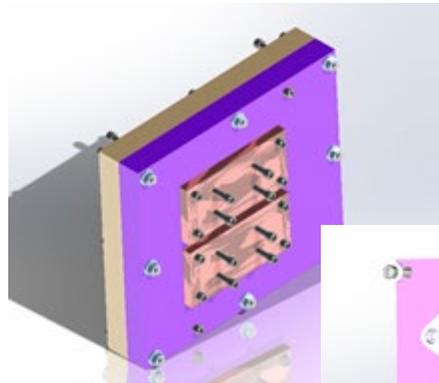


Lug Component in Tooling Mold



Lug Preform Showing Stitch Density Variations

- Building Complex Clip Bracket
  - Comparison to actual component cost and weight)



# Summary / Q&A

- The IACMI projects are providing essential building blocks to enable lighter and more cost effective structure
- The teamwork and collaboration within IACMI enables adoption of overmolding technology into many applications in the future

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# Thank you

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